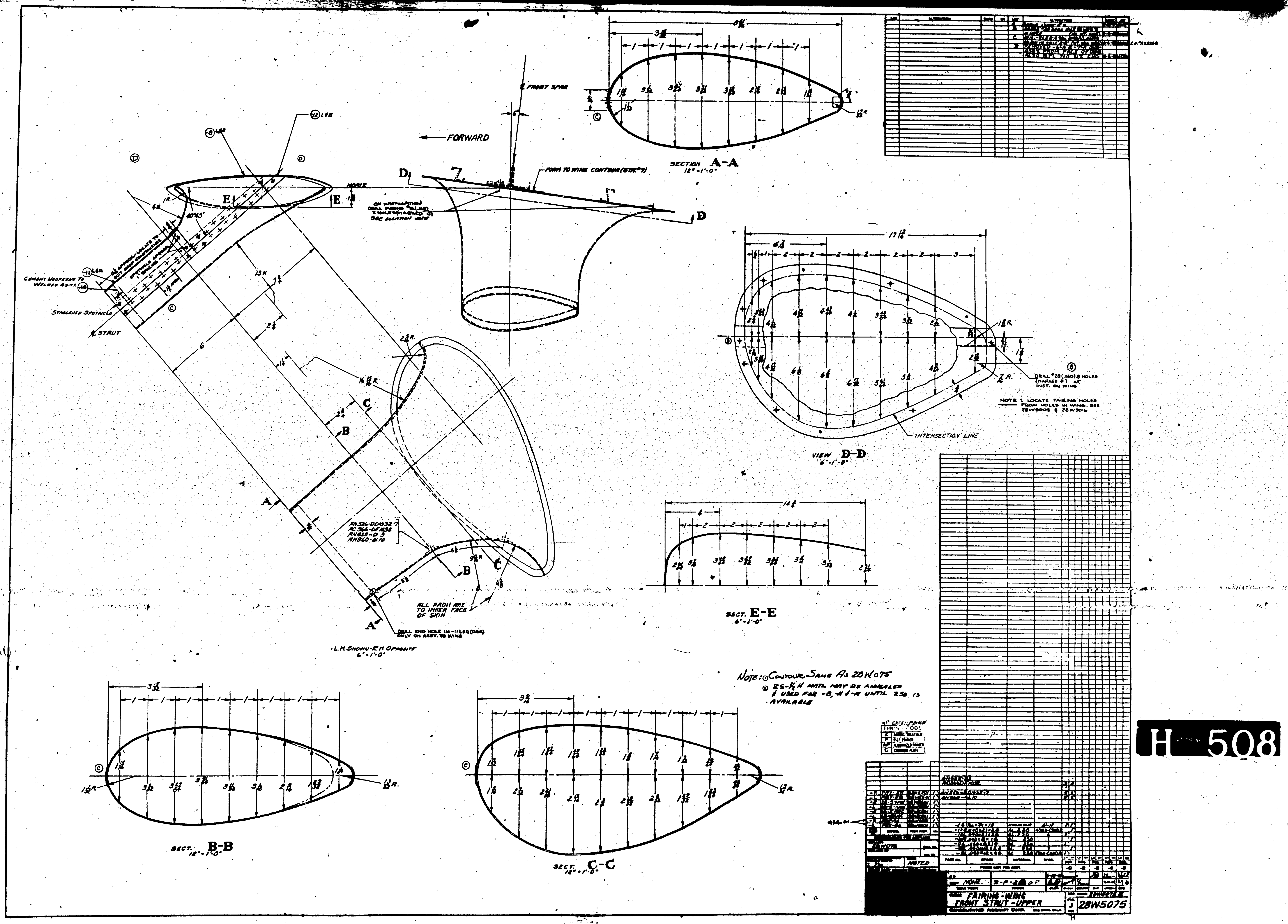
FORWARD
FRONT SPAR
FORM TO WING CONTOUR (STA. 7)
ON INSTALLATION DRILL ENDING #8 (.140) 8 HOLES (MARKED ◊) SEE LOCATION NOTE
CEMENT NEOPRENE TO WELDED ASSY.
STAGGERED SPOTWELD
STRUT
ALL RADII ARE TO INNER FACE OF SKIN
DRILL END HOLE IN -11 L&R (2EA) ONLY ON ASSY. TO WING
AN526-DD1032-7
AC364-DF1032
AN425-D3
AN960-Al10
L.H. SHOWN - R.H. OPPOSITE 6"=1'-0"
SECTION A-A 12"=1'-0"
VIEW D-D 6"=1'-0"
INTERSECTION LINE
DRILL #28 (.140) 8 HOLES (MARKED ◊) AT INST. ON WING
NOTE: LOCATE FAIRING HOLES FROM HOLES IN WING. SEE 28W5005 & 28W5016
SECT. E-E 6"=1'-0"
SECT. B-B 12"=1'-0"
SECT. C-C 12"=1'-0"
NOTE: ① CONTOUR SAME AS 28W075
② 2S-½H MATL MAY BE ANNEALED & USED FOR -8, -11 & -12 UNTIL 2SO IS AVAILABLE
-P GREEN PRIME
FINISH CODE
Z ANODIC TREATMENT
P P-27 PRIMER
AP ALUMINIZED PRIMER
C CADMIUM PLATE
PART NO. STOCK MATERIAL SPEC.
PARTS LIST FOR ASSY.
NOTED
NONE
FAIRING - WING FRONT STRUT - UPPER
CONSOLIDATED AIRCRAFT CORP.
J
28W5075
H 508